Dynamics of a Rolling Coin

Let us consider the motion of a coin rolling on a horizontal table without slipping. This "rolling without slipping" constraint cannot be expressed as a function of coordinates, and is called a non-holonomic constraint, which makes the coin's motion complex.

The parameters characterizing the system are the mass M and the moment of inertia tensor \hat{I} about the center of mass, the radius of the coin a, and the gravitational constant g.

1 Equations of Motion

With the center of mass velocity v and the angular velocity ω of the coin, the equations of motion are

$$M\frac{d\boldsymbol{v}}{dt} = M\boldsymbol{g} + \boldsymbol{R} \qquad \text{Translational motion} \qquad (1)$$
$$\frac{d}{dt} \left(\hat{I} \boldsymbol{\omega} \right) = \boldsymbol{c} \times \boldsymbol{R} \qquad \text{Rotational motion} \qquad (2)$$

These are two vector equations. Here, g is the gravitational acceleration vector of magnitude g pointing vertically downward, \mathbf{R} is the reaction force from the floor, and the vector \mathbf{c} points from the center of the coin to the contact point with the floor.

The constraint of rolling without slipping relates the center of mass velocity v and the angular velocity ω as follows:

$$\boldsymbol{v} = -\boldsymbol{\omega} \times \boldsymbol{c}.$$
 (no-slip condition) (3)

This cannot be expressed by the relation of coordinates and angles, so it is a non-holonomic constraint.

By eliminating \mathbf{R} from Eqs. (1) and (2), and using Eq. (3) to express \mathbf{v} in terms of $\boldsymbol{\omega}$, the equation of motion becomes

$$\frac{d}{dt}(\hat{I}\boldsymbol{\omega}) = M\boldsymbol{c} \times \left(-\frac{d}{dt}(\boldsymbol{\omega} \times \boldsymbol{c}) - \boldsymbol{g}\right).$$
(4)

2 Coordinate Systems

Here, we introduce two coordinate systems, xyz and XYZ.

- (i) xyz: a inertial coordinate system. The horizontal floor is the z-x plane, and the vertical upward direction is taken as the y-axis.
- (ii) XYZ: the origin at the center of the coin, with the Z-axis perpendicular to the coin, X-axis lies in the horizontal plane, and the Y-axis is upward in the coin plane.

The basis vectors, i.e., the unit vectors parallel to each axis, are denoted by

$$(\boldsymbol{e}_x, \, \boldsymbol{e}_y, \, \boldsymbol{e}_z), \qquad (\boldsymbol{e}_X, \, \boldsymbol{e}_Y, \, \boldsymbol{e}_Z),$$
 (5)

where \boldsymbol{e}_X and \boldsymbol{e}_Y are defined as

$$\boldsymbol{e}_X \equiv \frac{\boldsymbol{e}_y \times \boldsymbol{e}_Z}{|\boldsymbol{e}_y \times \boldsymbol{e}_Z|}, \quad \boldsymbol{e}_Y \equiv \boldsymbol{e}_Z \times \boldsymbol{e}_X.$$
 (6)

Furthermore, we define three angles (θ, ϕ, ψ) to represent the orientation of the coin. That is, when $(\theta, \phi, \psi) = 0$, the coin lies in the *xy*-plane, and the *XYZ* axes are parallel to the *xyz* axes. The coin in orientation (θ, ϕ, ψ) is obtained from the (0, 0, 0) orientation by

- 1. rotating by ψ around the z-axis,
- 2. rotating by θ around the x-axis,
- 3. rotating by ϕ around the *y*-axis,

in that order.



🖾 1: Two coordinate systems xyz and XYZ, and three angles (θ, ϕ, ψ) that represent the coin's orientation (left figure). View from the Z axis (center) and the -X axis (right). When $(\theta, \phi) = 0$, the two coordinate systems coincide.

Then, the relationship between basis vectors can be expressed by the rotation matrix

$$\hat{R}(\theta,\phi) \equiv \begin{pmatrix}
\cos\phi, & 0, & -\sin\phi \\
\sin\theta\sin\phi, & \cos\theta, & \sin\theta\cos\phi \\
\cos\theta\sin\phi, & -\sin\theta, & \cos\theta\cos\phi
\end{pmatrix} = \hat{R}(x,\theta)\hat{R}(y,\phi)$$
(7)
$$\hat{R}^{t}(\theta,\phi) = \begin{pmatrix}
\cos\phi, & \sin\theta\sin\phi, & \cos\theta\sin\phi \\
0, & \cos\theta, & -\sin\theta
\end{pmatrix} = \hat{R}^{-1}(\theta,\phi)$$
(8)

$$\left(-\sin\phi, \sin\theta\cos\phi, \cos\theta\cos\phi \right)$$

as follows:

$$\begin{pmatrix} \mathbf{e}_{X} \\ \mathbf{e}_{Y} \\ \mathbf{e}_{Z} \end{pmatrix} = \hat{R}(\theta, \phi) \begin{pmatrix} \mathbf{e}_{x} \\ \mathbf{e}_{y} \\ \mathbf{e}_{z} \end{pmatrix}, \qquad \begin{pmatrix} \mathbf{e}_{x} \\ \mathbf{e}_{y} \\ \mathbf{e}_{z} \end{pmatrix} = \hat{R}^{t}(\theta, \phi) \begin{pmatrix} \mathbf{e}_{X} \\ \mathbf{e}_{Y} \\ \mathbf{e}_{Z} \end{pmatrix}$$
(9)

Here, $\hat{R}(x,\theta)$ and $\hat{R}(y,\phi)$ represent the rotation matrices about the x- and y-axes by angles θ and ϕ , respectively, namely

$$\hat{R}(x,\theta) = \begin{pmatrix} 1, & 0, & 0\\ 0, & \cos\theta, & \sin\theta\\ 0, & -\sin\theta, & \cos\theta \end{pmatrix}, \quad \hat{R}(y,\phi) = \begin{pmatrix} \cos\phi, & 0, & -\sin\phi\\ 0, & 1, & 0\\ \sin\phi, & 0, & \cos\phi \end{pmatrix}.$$
(10)

Let Ω be the angular velocity vector of the XYZ coordinate system. Then, Ω is given by the angular velocity vector of the coin ω as

$$\boldsymbol{\Omega} = \dot{\phi} \boldsymbol{e}_y + \dot{\theta} \boldsymbol{e}_X, \qquad \boldsymbol{\omega} = \dot{\phi} \boldsymbol{e}_y + \dot{\theta} \boldsymbol{e}_X + \dot{\psi} \boldsymbol{e}_Z. \tag{11}$$

3 Equations of Motion in the *XYZ* System

We express the equations of motion in the basis of the XYZ coordinate system. For a vector \boldsymbol{u} , the time derivative of its components in the XYZ frame is defined as

$$\left(\frac{d\boldsymbol{u}}{dt}\right)_{XYZ} \equiv \frac{du_X}{dt}\boldsymbol{e}_X + \frac{du_Y}{dt}\boldsymbol{e}_Y + \frac{du_Z}{dt}\boldsymbol{e}_Z.$$

Since the XYZ frame rotates with angular velocity Ω , the time derivative of u is given by

$$\frac{d\boldsymbol{u}}{dt} = \left(\frac{d\boldsymbol{u}}{dt}\right)_{XYZ} + \boldsymbol{\Omega} \times \boldsymbol{u}.$$

In the XYZ system, the inertia tensor is given by

$$\hat{I} = \begin{pmatrix} \frac{1}{4}Ma^2, & 0, & 0\\ 0, & \frac{1}{4}Ma^2, & 0\\ 0, & 0, & \frac{1}{2}Ma^2 \end{pmatrix}_{XYZ}$$
(12)

The contact vector \boldsymbol{c} and the gravitational acceleration vector \boldsymbol{g} are

$$\boldsymbol{c} = -a\boldsymbol{e}_Y, \qquad \boldsymbol{g} = -g\boldsymbol{e}_y.$$

Therefore, the equation of motion (4) becomes

$$\hat{I}\left(\frac{d\boldsymbol{\omega}}{dt}\right)_{XYZ} + \boldsymbol{\Omega} \times \left(\hat{I}\boldsymbol{\omega}\right) = M\boldsymbol{c} \times \left(-\left(\frac{d\boldsymbol{\omega}}{dt}\right)_{XYZ} \times \boldsymbol{c} + \boldsymbol{\Omega} \times \left(-\boldsymbol{\omega} \times \boldsymbol{c}\right) + g\boldsymbol{e}_{y}\right).$$

Rearranging this, we obtain

$$\hat{I}\left(\frac{d\boldsymbol{\omega}}{dt}\right)_{XYZ} + Ma^2\left(\left(\frac{d\boldsymbol{\omega}}{dt}\right)_{XYZ} - \frac{d\omega_Y}{dt}\boldsymbol{e}_Y\right) = -\boldsymbol{\Omega} \times \left(\hat{I}\boldsymbol{\omega}\right) - Ma^2\left(\Omega_Y\boldsymbol{\omega} \times \boldsymbol{e}_Y + \frac{g}{a}\boldsymbol{e}_Y \times \boldsymbol{e}_y\right).$$

The angular velocity $\pmb{\omega}$ is related with the angular velocity of the XYZ coordinate system $\pmb{\Omega}$ as

$$\boldsymbol{\omega} = \boldsymbol{\Omega} + \dot{\psi} \boldsymbol{e}_Z; \quad \text{i.e.} \quad \omega_X = \Omega_X, \quad \omega_Y = \Omega_Y, \quad \omega_Z = \Omega_Z + \dot{\psi}. \quad (13)$$

Expanding the components of the equation of motion in the XYZ coordinate system, we obtain

$$(I_X + Ma^2)\frac{d\omega_X}{dt} = -(I_Z + Ma^2)\omega_Y\omega_Z - I_X\omega_Y^2\tan\theta + Ma^2\frac{g}{a}\sin\theta$$
(14)

$$I_X \frac{d\omega_Y}{dt} = +I_X \omega_X \omega_Y \tan \theta + I_Z \omega_Z \omega_X \tag{15}$$

$$(I_Z + Ma^2)\frac{d\omega_Z}{dt} = +Ma^2\omega_X\omega_Y \tag{16}$$

Here, $I_X = \frac{1}{4}Ma^2$, $I_Z = \frac{1}{2}Ma^2$. From Eq. (11), the angular velocity $\boldsymbol{\omega}$ can be expressed by the orientation angles (θ, ϕ, ψ) as

$$\omega_X = \dot{\theta}, \qquad \omega_Y = \dot{\phi} \cos \theta, \qquad \omega_Z = \dot{\psi} - \dot{\phi} \sin \theta,$$

from which we obtain

$$d\theta$$
 (17)

$$\frac{dt}{dt} = \omega_X,\tag{11}$$

$$\frac{d\phi}{dt} = \frac{1}{\cos\theta}\,\omega_Y,\tag{18}$$

$$\frac{d\psi}{dt} = \omega_Z + \tan\theta\,\omega_Y.\tag{19}$$

4 Physical Quantities

Center of Mass Velocity and Position:

$$\boldsymbol{v} = -\boldsymbol{\omega} \times (-a\boldsymbol{e}_Y) = a(-\omega_Z \boldsymbol{e}_X + \omega_X \boldsymbol{e}_Z)$$
$$\dot{\boldsymbol{x}} = v_x = a(-\omega_Z \cos\phi + \omega_X \cos\theta \sin\phi)$$
$$\dot{\boldsymbol{y}} = v_y = -a\omega_X \sin\theta$$
$$\dot{\boldsymbol{z}} = v_z = a(\omega_Z \sin\phi + \omega_X \cos\theta \cos\phi)$$

Contact Point with Floor:

$$\mathbf{r}_{c} = \mathbf{r} + \mathbf{c}; \qquad x_{c} = x - a\sin\theta\sin\phi, \quad z_{c} = z - a\sin\theta\cos\phi$$

$$\begin{cases} \dot{x}_{c} = -a(\tan\theta\omega_{Y} + \omega_{Z})\cos\phi = -a\dot{\psi}\cos\phi \\ \dot{z}_{c} = +a(\tan\theta\omega_{Y} + \omega_{Z})\sin\phi = +a\dot{\psi}\sin\phi \end{cases} \Rightarrow \quad v_{c} = a|\dot{\psi}|$$

Energy:

$$E = \frac{1}{2}Mv^2 + \frac{1}{2}\boldsymbol{\omega}^t \hat{I}\boldsymbol{\omega} + Mg(-\boldsymbol{c}) \cdot \boldsymbol{e}_y = \frac{1}{2}Mv^2 + \frac{1}{2}\boldsymbol{\omega}^t \hat{I}\boldsymbol{\omega} + Mga\cos\theta$$
$$= \frac{1}{2}Ma^2(\omega_X^2 + \omega_Z^2) + \frac{1}{2}(I_X(\omega_X^2 + \omega_Y^2) + I_Z\omega_Z^2) + Mga\cos\theta$$

Reaction Force:

$$\mathbf{R} = M \frac{d\mathbf{v}}{dt} + Mg\mathbf{e}_y = Ma \frac{d}{dt} (\mathbf{\omega} \times \mathbf{e}_Y) + Mg\mathbf{e}_y$$

= $Ma \Big(-(\dot{\omega}_Z - \omega_X \omega_Y)\mathbf{e}_X - (\omega_X^2 - \omega_Y \omega_Z \tan \theta)\mathbf{e}_Y + (\dot{\omega}_X + \omega_Y \omega_Z)\mathbf{e}_Z \Big) + Mg\mathbf{e}_y$

$$R_{x} = Ma \Big(-(\dot{\omega}_{Z} - \omega_{X}\omega_{Y})\cos\phi - (\omega_{X}^{2} - \omega_{Y}\omega_{Z}\tan\theta)\sin\theta\sin\phi + (\dot{\omega}_{X} + \omega_{Y}\omega_{Z})\cos\theta\sin\phi \Big)$$
$$R_{y} = Ma \Big(-(\omega_{X}^{2} - \omega_{Y}\omega_{Z}\tan\theta)\cos\theta - (\dot{\omega}_{X} + \omega_{Y}\omega_{Z})\sin\theta \Big) + Mg$$
$$R_{z} = Ma \Big((\dot{\omega}_{Z} - \omega_{X}\omega_{Y})\sin\phi - (\omega_{X}^{2} - \omega_{Y}\omega_{Z}\tan\theta)\sin\theta\cos\phi + (\dot{\omega}_{X} + \omega_{Y}\omega_{Z})\cos\theta\cos\phi \Big)$$

$$F \equiv \sqrt{R_x^2 + R_y^2} = Ma \sqrt{\left(\dot{\omega}_Z - \omega_X \omega_Y\right)^2 + \left(\omega_X^2 \sin \theta - \dot{\omega}_Z \cos \theta - \frac{\omega_Y \omega_Z}{\cos \theta}\right)^2}$$
$$N \equiv R_y = -Ma \left(\omega_X^2 \cos \theta + \dot{\omega}_X \sin \theta\right) + Mg$$

Coin Orientation: When $(\theta, \phi, \psi) = 0$, the coin lies in the *x*-*z* plane and the *Z*-axis points opposite direction to the *y*-axis. To orient the coin to (θ, ϕ, ψ) from this state, the following rotations should be performed in this sequence:

- 1. rotating by ψ around the z-axis,
- 2. rotating by θ around the *x*-axis,
- 3. rotating by ϕ around the *y*-axis,

The rotation operator \hat{R} of quaternion for a rotation of angle θ about axis \boldsymbol{n} is

$$\hat{R}(\boldsymbol{n},\theta) = \cos\frac{\theta}{2} + \hat{n}\sin\frac{\theta}{2}; \quad \hat{n} \equiv n_x\hat{i} + n_y\hat{j} + n_z\hat{k}$$

Therefore, \hat{R} is given by

$$\hat{R} = \hat{R}(\boldsymbol{e}_y, \phi) \hat{R}(\boldsymbol{e}_x, \theta) \hat{R}(\boldsymbol{e}_z, \psi),$$
$$\hat{R}(\boldsymbol{e}_z, \psi) = \cos \frac{\psi}{2} + \hat{k} \sin \frac{\psi}{2},$$
$$\hat{R}(\boldsymbol{e}_x, \theta) = \cos \frac{\theta}{2} + \hat{i} \sin \frac{\theta}{2},$$
$$\hat{R}(\boldsymbol{e}_y, \phi) = \cos \frac{\phi}{2} + \hat{j} \sin \frac{\phi}{2}.$$

A quaternion is an extension of complex numbers with the units \hat{i} , \hat{j} , \hat{k} , and 1, which follow the multiplication rules

$$\hat{i}^2 = \hat{j}^2 = \hat{k}^2 = -1, \qquad \hat{i}\,\hat{j} = -\hat{j}\,\hat{i} = \hat{k}, \quad \hat{j}\,\hat{k} = -\hat{k}\,\hat{j} = \hat{i}, \quad \hat{k}\,\hat{i} = -\hat{i}\,\hat{k} = \hat{j}.$$

5 Steady States and Stability Analysis

We look into steady solutions with the coenstant angular momentum, and analyze their stability. To simplify the equations, we adopt a dimensionless unit system with M = a = g = 1, where

$$I_X = I_Y = \frac{1}{4}, \quad I_Z = \frac{1}{2},$$

and the equations of motion become

The steady solutions satisfy

$$\begin{cases} 0 = -\frac{6}{5}\omega_Y\omega_Z - \frac{1}{5}\omega_Y^2\tan\theta + \frac{4}{5}\sin\theta \\ 0 = \omega_X\omega_Y\tan\theta + 2\omega_Z\omega_X \\ 0 = \frac{2}{3}\omega_X\omega_Y \end{cases},$$

from which, we obtain two conditions:

1.
$$\omega_X = \omega_Y = \theta = 0$$
, $\omega_Z = \text{ const.}$

2.
$$\omega_X = 0$$
, $(\omega_Z, \theta) = \text{ const.}$, $\omega_Y = f^{\pm}(\omega_Z, \theta) \equiv \frac{1}{\tan \theta} \left(-3\omega_Z \pm \sqrt{9\omega_Z^2 + 4\frac{\sin^2 \theta}{\cos \theta}} \right)$

The function $f^{\pm}(\omega_Z, \theta)$ has

$$f^{\pm}(\omega_Z,\theta) = -f^{\mp}(-\omega_Z,\theta) = f^{\mp}(-\omega_Z,-\theta)$$

symmetry, so we may consider only $\omega_Z, \theta > 0$.

Steady State 1: Straight Rolling Motion

$$\omega_Z = \omega_{Z,0} = \text{const.}, \quad \omega_X = \omega_Y = \theta = 0$$

Steady State 2: Spinning with Stationary Center of Mass, and F = 0, R = -Mg

$$\omega_X = \omega_Z = 0, \quad \omega_Y = 2\sqrt{\cos\theta_0}, \quad \theta = \theta_0, \qquad \dot{\phi} = \frac{2}{\sqrt{\cos\theta_0}}$$

Radius of the contact point trajectory: R_C

$$R_C = \frac{\dot{\psi}}{\dot{\phi}} = \sin\theta_0$$

Steady State 3: Circular Motion

$$\omega_X = 0, \ \omega_Z = \omega_{Z,0}, \ \theta = \theta_0, \quad \omega_Y = \omega_{Y,0} \equiv \frac{3\omega_{Z,0}}{\tan\theta_0} \left(-1 \pm \sqrt{1 + \frac{4}{9} \frac{\tan\theta_0}{\omega_{Z,0}^2} \sin\theta_0} \right)$$

Radius of COM trajectory: R_G ,

$$R_G = \left|\frac{\omega_Z}{\dot{\phi}}\right| = \left|\frac{\omega_{Z,0}}{\omega_{Y,0}}\cos\theta_0\right| = \frac{3}{4}\frac{\omega_{Z,0}^2}{\tan\theta_0}\left(\sqrt{1 + \frac{4}{9}\frac{\tan\theta_0}{\omega_{Z,0}^2}}\sin\theta_0 \pm 1\right)$$

Radius of the contact point trajectory: R_C

$$R_C = \left| \frac{\dot{\psi}}{\dot{\phi}} \right| = \left| \frac{\omega_{Z,0}}{\omega_{Y,0}} \cos \theta_0 + \sin \theta_0 \right|$$
$$= \left| \frac{3}{4} \frac{\omega_{Z,0}^2}{\tan \theta_0} \left(\sqrt{1 + \frac{4}{9} \frac{\tan \theta_0}{\omega_{Z,0}^2} \sin \theta_0} \pm 1 \right) \pm \sin \theta_0 \right|$$

Steady State 4: Large Circular Motion

$$\omega_X = 0, \ \omega_Z = \omega_{Z,0}, \ \theta = \theta_0 \ll 1$$

$$\omega_Y = \omega_{Y,0} \equiv \frac{3\omega_{Z,0}}{\tan\theta_0} \left(\sqrt{1 + \frac{4}{9} \frac{\tan\theta_0}{\omega_{Z,0}^2}} \sin\theta_0 - 1 \right) \approx \frac{2}{3} \frac{\sin\theta_0}{\omega_{Z,0}}$$
$$R_G = \frac{3}{4} \frac{\omega_{Z,0}^2}{\tan\theta_0} \left(\sqrt{1 + \frac{4}{9} \frac{\tan\theta_0}{\omega_{Z,0}^2}} \sin\theta_0 + 1 \right) \approx \frac{3}{2} \frac{\omega_{Z,0}^2}{\sin\theta_0} \approx \frac{\omega_{Z,0}}{\omega_{Y,0}}$$

5.1 Stability of Steady State 1:

 $\omega_Z = \omega_{Z,0} = \text{const.}, \quad \omega_X = \omega_Y = \theta = 0$

Let small deviations from this state be

$$\omega_X = \delta \omega_X, \quad \omega_Y = \delta \omega_Y, \quad \omega_Z = \omega_{Z,0} + \delta \omega_Z, \quad \theta = \delta \theta$$

Then,

,

$$\begin{cases} \delta\dot{\omega}_X = -\frac{6}{5}\omega_{Z,0}\delta\omega_Y + \frac{4}{5}\delta\theta \\ \delta\dot{\omega}_Y = 2\omega_{Z,0}\delta\omega_X \\ \delta\dot{\omega}_Z = 0 \end{cases} \qquad \begin{cases} \delta\dot{\theta} = \delta\omega_X \\ \delta\dot{\phi} = \delta\omega_Y \\ \dot{\psi} = \omega_{Z,0} \end{cases}$$

Assuming $(\delta \omega_X, \delta \omega_Y, \delta \theta) = (A_X, A_Y, A_\theta) e^{\lambda t}$, we have

$$\lambda \boldsymbol{A} = \hat{\Lambda} \boldsymbol{A}; \qquad \hat{\Lambda} \equiv \begin{pmatrix} 0, & -\frac{6}{5}\omega_{Z,0}, & \frac{4}{5}\\ 2\omega_{Z,0}, & 0, & 0\\ 1, & 0, & 0 \end{pmatrix}$$

Characteristic equation:

$$\lambda \left(\lambda^2 + \frac{12}{5} \omega_{Z,0}^2 - \frac{4}{5} \right) = 0$$

Eigenvalues:

$$\lambda = 0, \quad \pm \sqrt{\frac{4}{5} \left(1 - 3\omega_{Z,0}^2\right)},$$

from which the critical angular velocity is $\omega_c \equiv 1/\sqrt{3} \approx 0.577$.

Stability condition: Unstable if $\omega_{Z,0} < \omega_c$, and neutrally stable if $\omega_{Z,0} \ge \omega_c$.

Neutral stability means that orbits of arbitrary radius exist, but with no restoring force, so even small perturbations result in transition to another circular trajectory.

5.2 Stability of Steady State 2:

Assume constants $\omega_X = 0$, and $(\omega_Y, \omega_Z, \theta) \neq 0$

$$I_X \omega_Y^2 \tan \theta + (I_Z + Ma^2) \omega_Y \omega_Z - Ma^2 \frac{g}{a} \sin \theta = 0$$

satisfies the condition. Let the solution be:

$$\omega_X = \delta \omega_X, \quad \omega_Y = \omega_{Y,0} + \delta \omega_X, \quad \omega_Z = \omega_{Z,0} + \delta \omega_X, \quad \theta = \theta_0 + \delta \theta$$

Assuming:

6 Case of a Ring

Moments of inertia of a ring with outer radius a and inner radius b:

$$I_X = I_Y = \frac{1}{4}M(a^2 + b^2), \qquad I_Z = \frac{1}{2}M(a^2 + b^2)$$